

A = DASH NUMBER (TABLE WIDTH)

NOTES:

TABLE TOP INFORMATION

A) THE COMPLIANCE OF THE INSTALLATION OF THE TABLE TOP TO FEDERAL REGULATION IS THE RESPONSIBILITY OF THE INSTALLER.

B) THE SIZE AND LOCATION OF THE TABLE TOP MOUNTING HOLES ARE AS INDICATED.

AIRCRAFT ATTACHMENT INFORMATION

A) STRUCTURAL SUBSTANTIATION OF THE AIRCRAFT STRUCTURE IS THE RESPONSIBILITY OF THE INSTALLER.

B) FASTENERS MUST NOT INTERFERE WITH THE OPERATION OF THE MECHANISM.

C) COMPLIANCE TO APPLICABLE FEDERAL AVIATION REGULATIONS REGARDING AISLE CLEARANCE REQUIREMENTS IS THE RESPONSIBILITY OF THE INSTALLER.

MECHANICAL INFORMATION


A) WEIGHT OF THE MECHANISM: 5.10 LBS FOR A 20" TABLE (+/- .04 LB PER INCH).

B) THE LOCATION OF THE CENTER OF GRAVITY IN THE STOWED POSITION IS AT X=-1.25, Y=6.82, Z=-12.74. THE CENTER OF GRAVITY IN THE DEPLOYED POSITION IS LOCATED AT X=-1.75, Y=10.80, Z=-12.75.

C) THE MECHANISM CAN SUPPORT AN ULTIMATE LOAD OF 120 LBS APPLIED 19.28 INCHES FROM THE INBOARD FACE OF THE SIDEPLATES AND CENTERED ON THE A DIMENSION. SEE STRESS REPORT R54800. CONVERT ULTIMATE LOAD TO AN ALLOWABLE TABLE TOP WEIGHT WITH THE FOLLOWING FORMULA: (WEIGHT OF ITEMS + WEIGHT OF TABLE TOP) X ULTIMATE LOAD FACTOR MUST BE LESS THAN OR EQUAL TO 120 LBS.

D) SEE OPERATION MANUAL M54800 FOR ADDITIONAL INFORMATION ON MECHANISM OPERATION AND MAINTENANCE.

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INSPECTION CLASS C		DRAWN: VVN	DATE: 12/14/2007	 Georgetown, Texas	
UNLESS OTHERWISE SPECIFIED		FINISH: N/A			
1) ALL SURFACES PER EPS-121-01-D		MATERIAL: N/A		TITLE	
2) REMOVE ALL BURRS & SHARP EDGES .010-.015				SIDE WALL TABLE 54800	
3) DIMENSIONS ARE IN INCHES		<i>The data, concepts or designs contained herein are the exclusive property of Enlite, Inc. or contain the proprietary rights of others and shall not be used, disclosed, copied, duplicated or reproduced without the written consent of Enlite, Inc.</i>		INSTALLATION DRAWING	
4) DO NOT SCALE DRAWING				SIZE	DWG NO.
5) TOLERANCES:		A	N54800	A	1 OF 1
DECIMAL .XX ± .04					
ANGULAR .XXX ± .01					
6) ALL MACHINED SURFACES 63/					